



Jet Propulsion Laboratory
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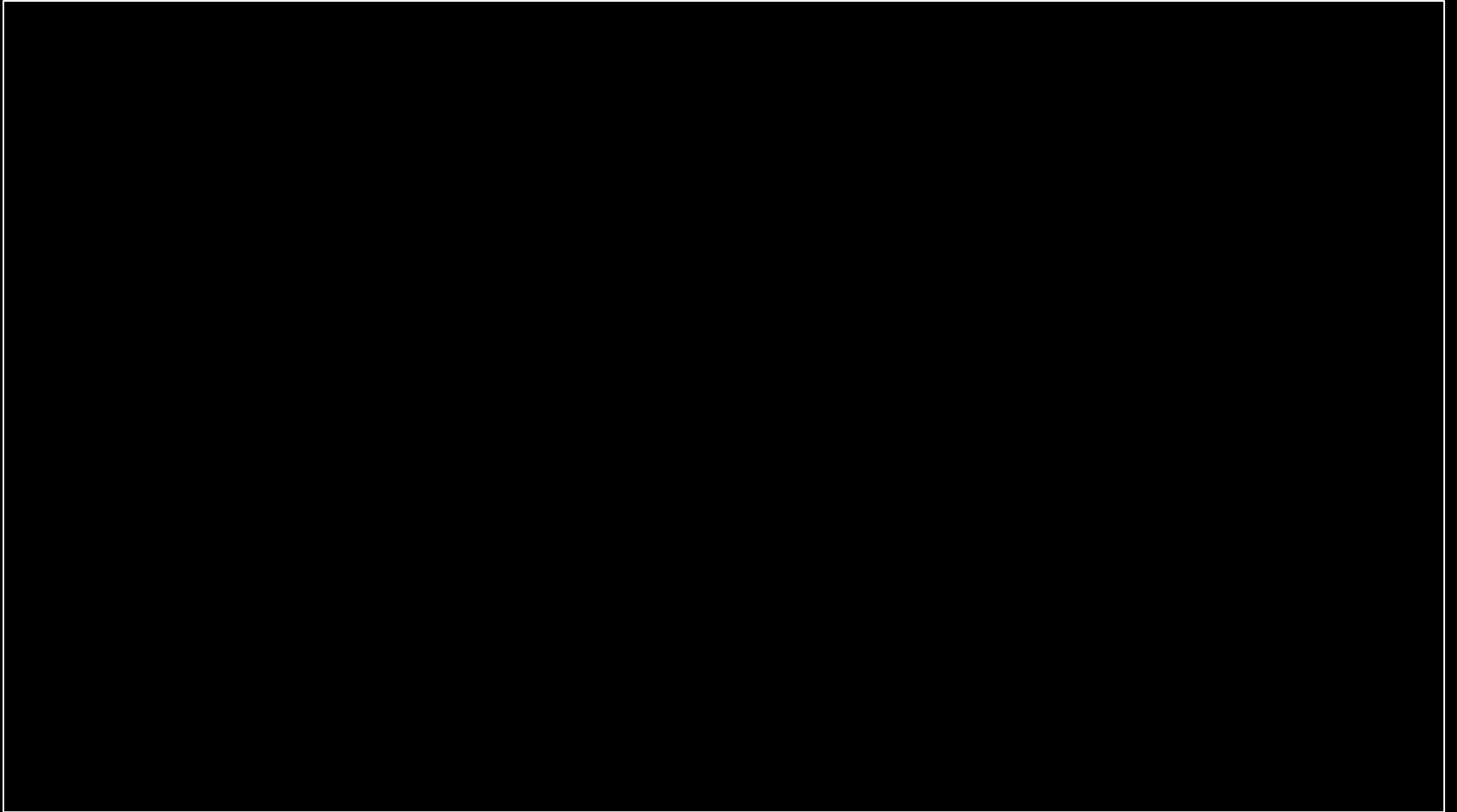


Mars Helicopter

Leveraging Commercial Hardware and Open Source Software

Timothy Canham

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NASA Mars Helicopter Page:

<https://www.jpl.nasa.gov/news/news.php?feature=7121>

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Conditions at Mars

Long distance from Earth

Thin atmosphere (<1% of Earth's)

Cold Martian nights (~ -90°C)

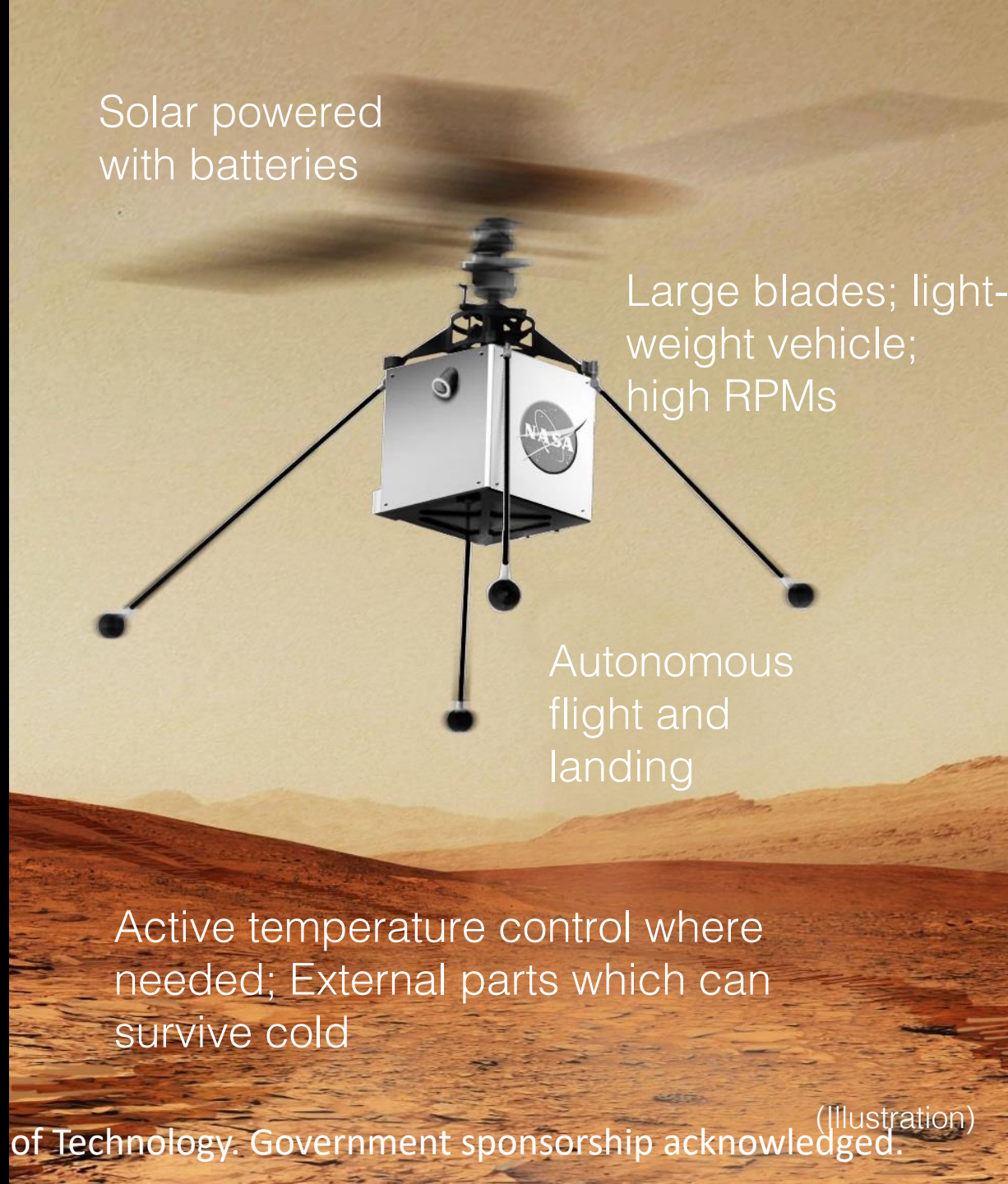
Need self-sufficient solar power system

Solar powered with batteries

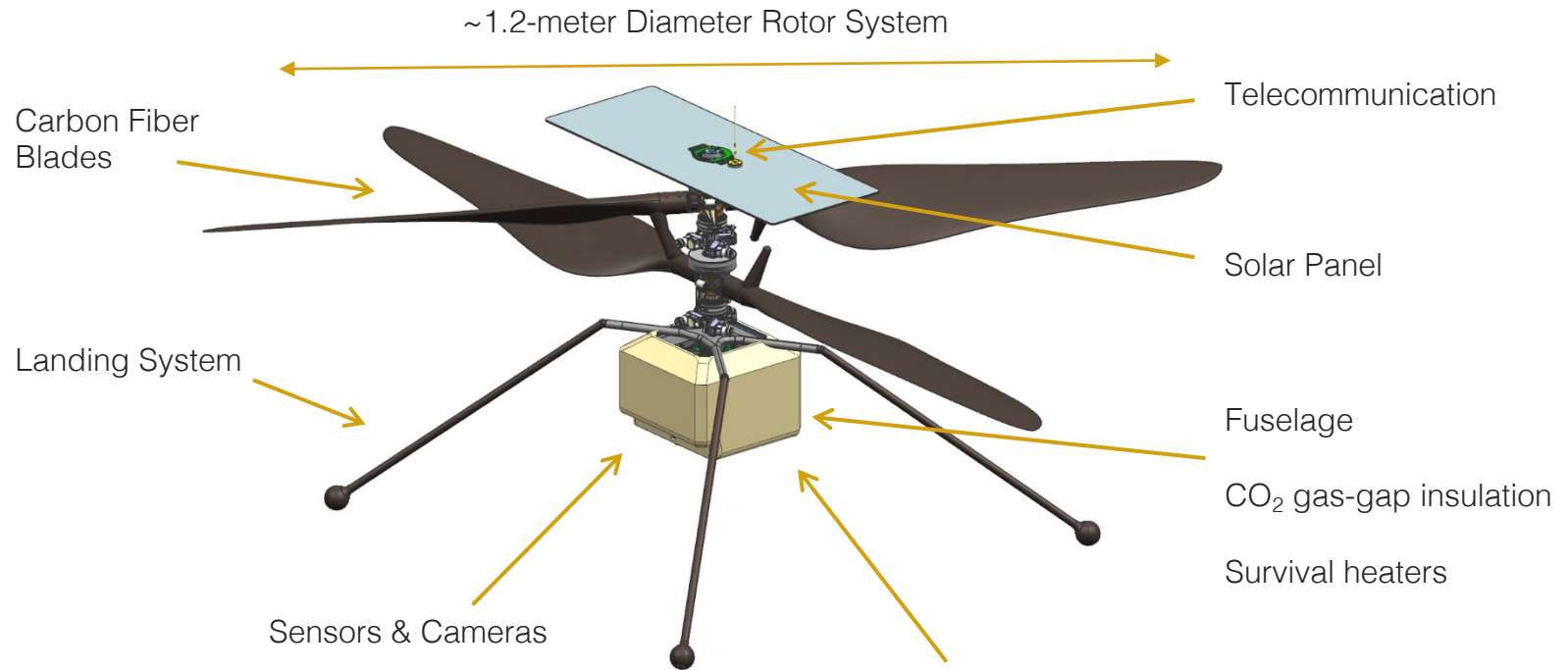
Large blades; lightweight vehicle; high RPMs

Autonomous flight and landing

Active temperature control where needed; External parts which can survive cold



Anatomy of Mars Helicopter



- ✓ Total Mass < 1.8 Kg
- ✓ Rotor Speed: 1900-2800 RPM
- ✓ Blade Tip Mach Number: < 0.7

Avionics Processing

Batteries

Autonomous flight control algorithms

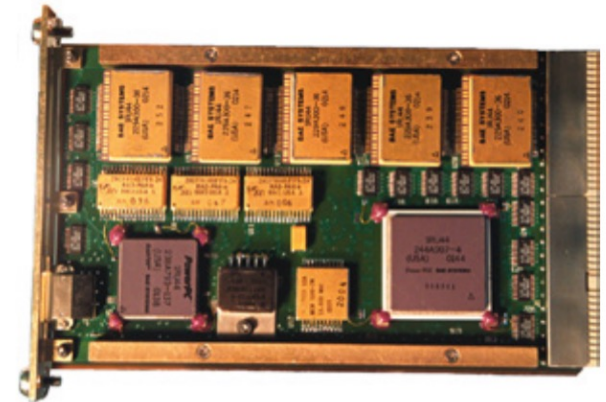
Built and Tested “Flight Model” Mars Helicopter



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Mars Helicopter Avionics Design

- The helicopter avionics needs to be light-weight, powerful, and low power.
 - 500Hz guidance loops
 - 30Hz vision-based navigation
- Current radiation-tolerant hardware is too bulky and does not have computing power needed
 - BAE RAD750 - ~200 DMIPS
- Choice was made to mix Commercial Off The Shelf (COTS) parts with some core radiation tolerant logic
 - RAD-Hard FPGA is the cop of the system
 - Provides clocks, core power management and watchdogs
 - Interfaces with sensors and motor system
 - Automotive grade microcontroller has responsibility for maintaining flight control (FC)
 - Fast microcontroller (~400DMIPS)
 - Dual lockstep processor can detect hardware faults
 - Redundant copies can fail over
 - Cell-phone grade ARM Linux processor does navigation, telecom, imaging and command/telemetry processing (NAV)
 - Very fast (~30,000 DMIPS)
 - Lots of memory
 - Not as robust as other parts
 - Cell-phone grade cameras for navigation and pictures
 - VGA gray-scale navigation
 - 13MP color camera for pictures



BAE RAD750

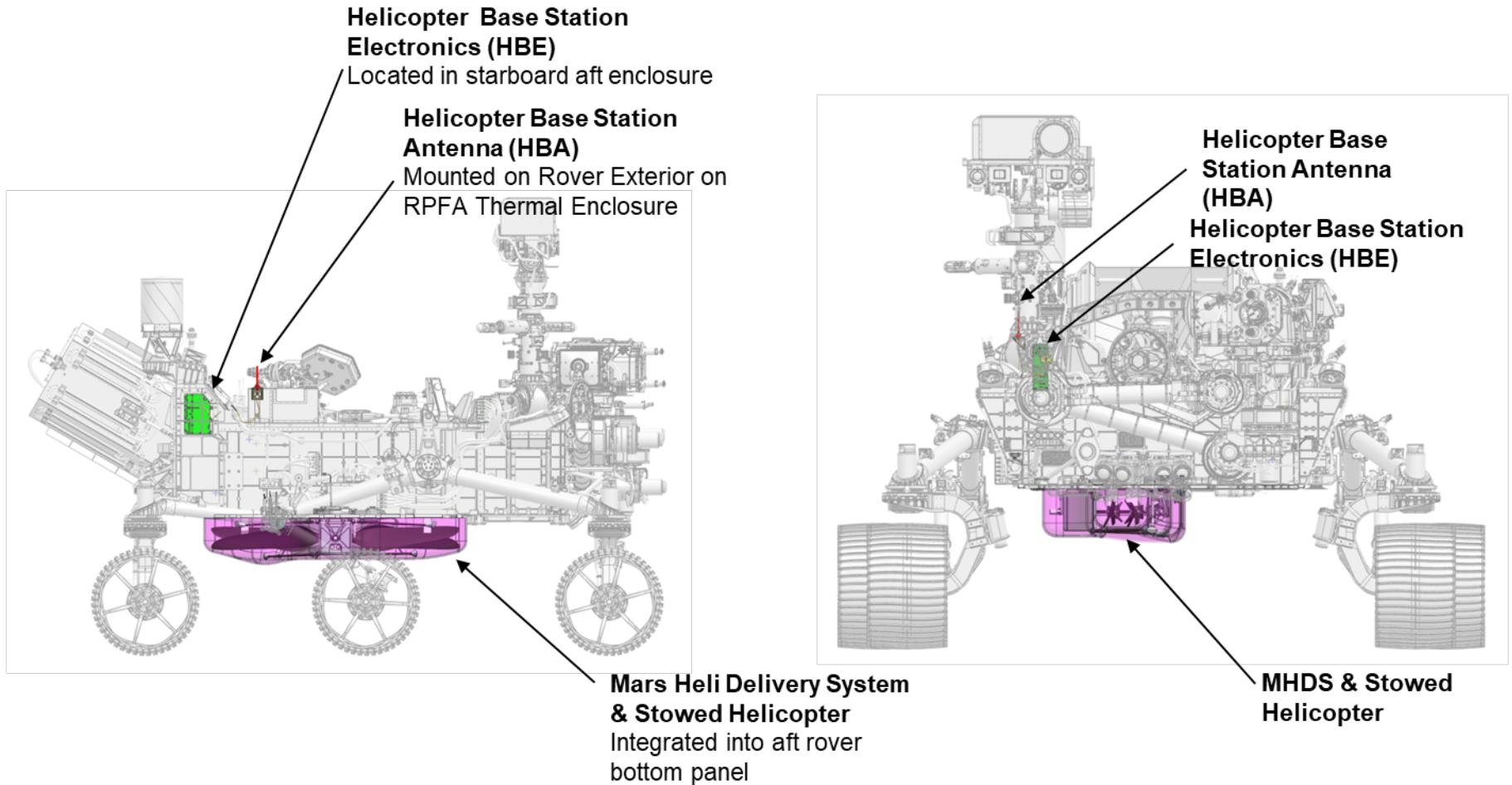


Texas Instruments TMS570



Snapdragon 801

Helicopter Elements on Rover



Mars Helicopter Avionics Design

■ Two major elements to the system:

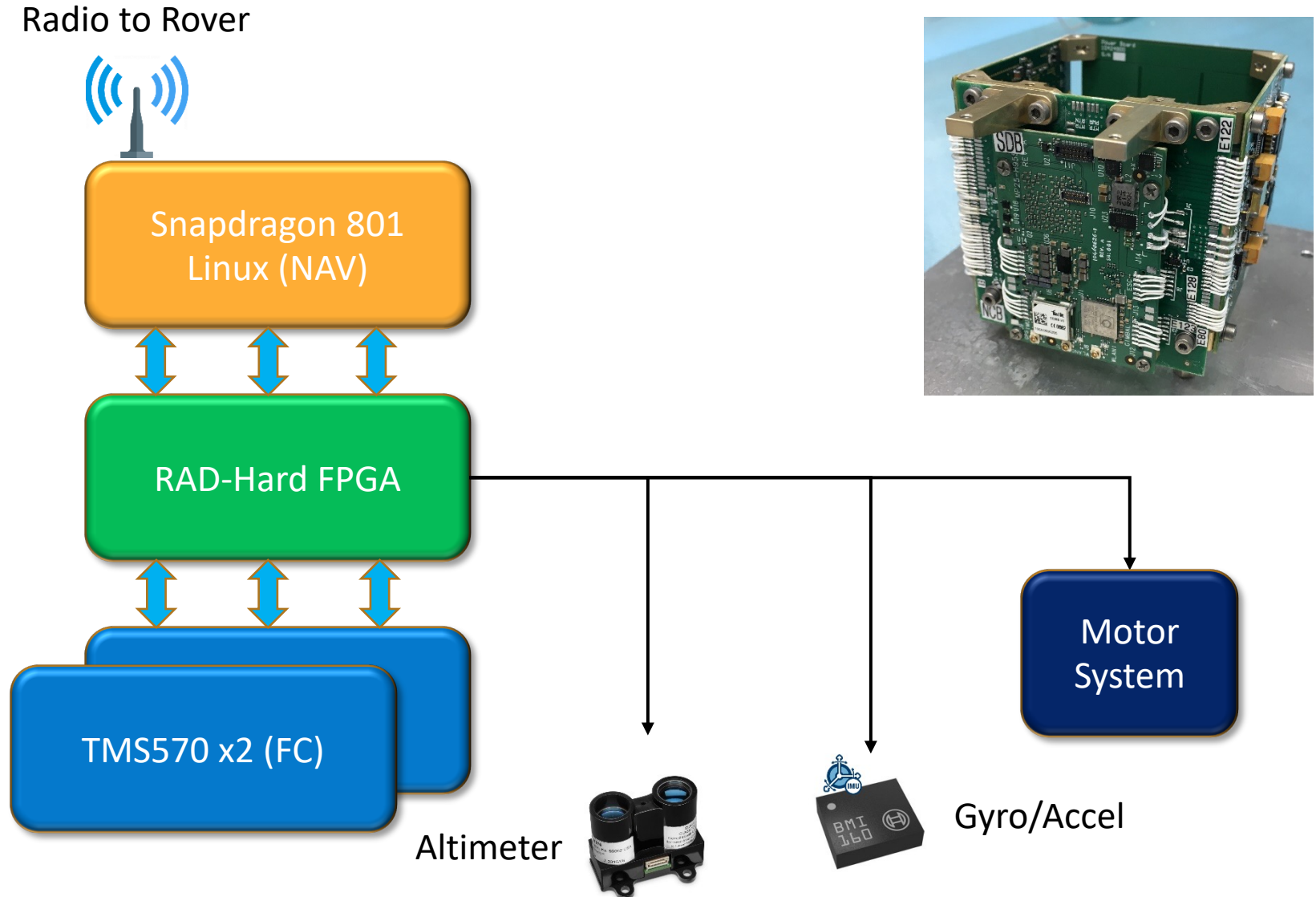
■ Helicopter

- Flying vehicle
- Solar-powered
- 1-2 minute flights
- NAV processor does command, telemetry, and radio functions, power/thermal management, feature tracking and “outer” guidance loop
 - 500Hz guidance, 30Hz tracking
 - Linux OS
- FC processor does “inner” guidance loop, flight attitude control, motor control and high-rate telemetry
 - 500Hz guidance and control
 - “Bare Metal”, no OS

■ Helicopter Base Station

- Permanently installed on rover
- Communicates with helicopter via radio after deployment
- Communicates with rover processor for commanding and data
- Performs battery charging prior to helicopter deployment

Mars Helicopter Avionics Design Block Diagram



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Mars Helicopter Base Station (HBS) Avionics Design Block Diagram

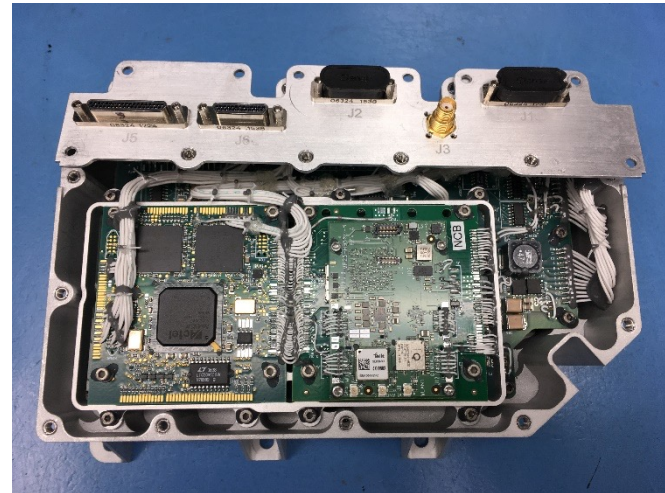
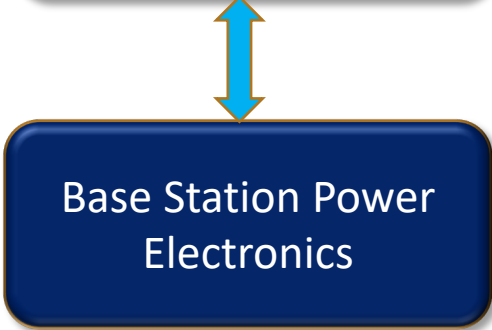
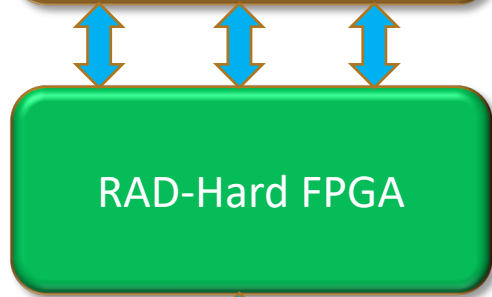
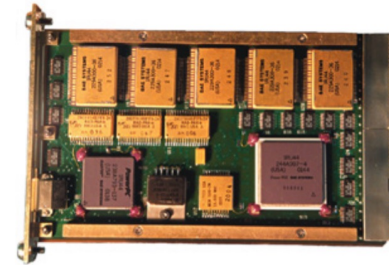
Radio to Mars Helicopter



RS-422 UART x2



Rover Processor



Mars Helicopter Flight Software

■ Smaller software team, shorter schedule

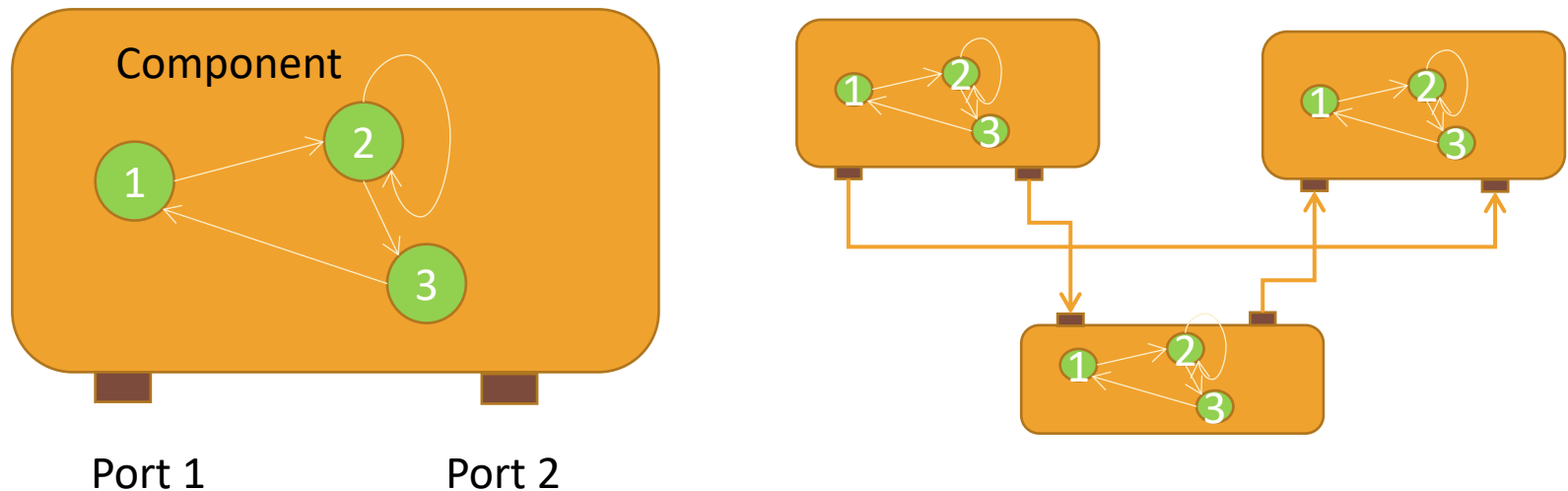
- No time to write software from scratch
- Needed a reliable code base
- Flexible architecture for multiple test configurations
 - Different test configurations for different venues

■ Chose F Prime flight software architecture

- JPL developed, but open sourced on NASA GitHub
 - <https://github.com/nasa/fprime>
- Used on previous projects at JPL
 - RapidScat
 - Asteria
- Planned future missions
 - NeaScout/Lunar Flashlight
 - Future smallsat/instrument Leon4 platform
- Collaboration with university CubeSat projects
- Helicopter reused many infrastructure components from previous projects

F Prime Architecture

- F Prime is a component architecture
- Software is composed of **components** (behaviors) and **ports** (interfaces between components)
- Components are interconnect together to form **topologies**, which comprise the binary built as a **deployment**.
- Components are not link dependent on other components, so they can be easily recombined to form alternate topologies.
- Mars Helicopter had 11 different deployments for various test venues and ground support applications.



F Prime Architecture

■ Serializables

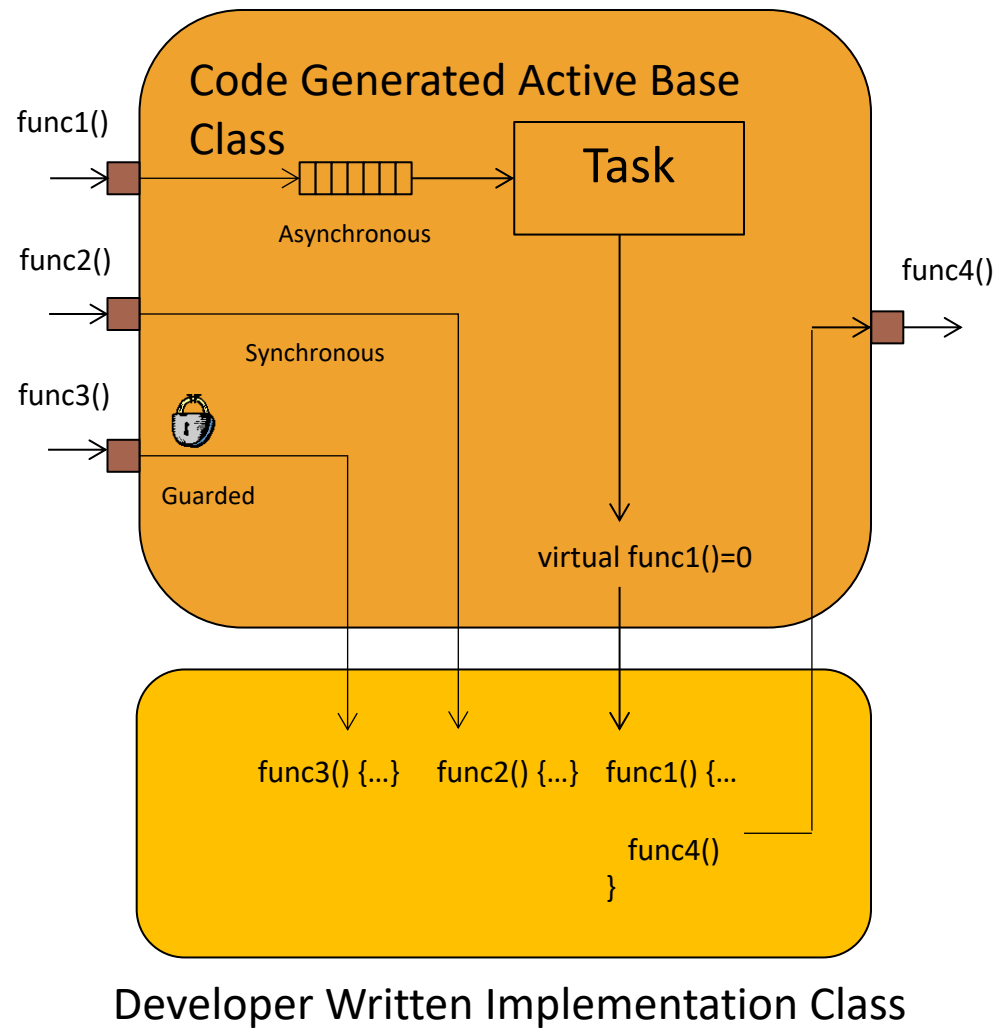
- Data types to pass between components

■ Port attributes

- Interface type
- Types passed to callee
- Optional types returned

■ Component Attributes

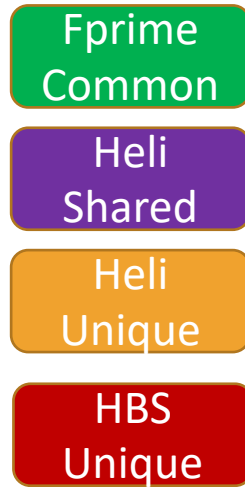
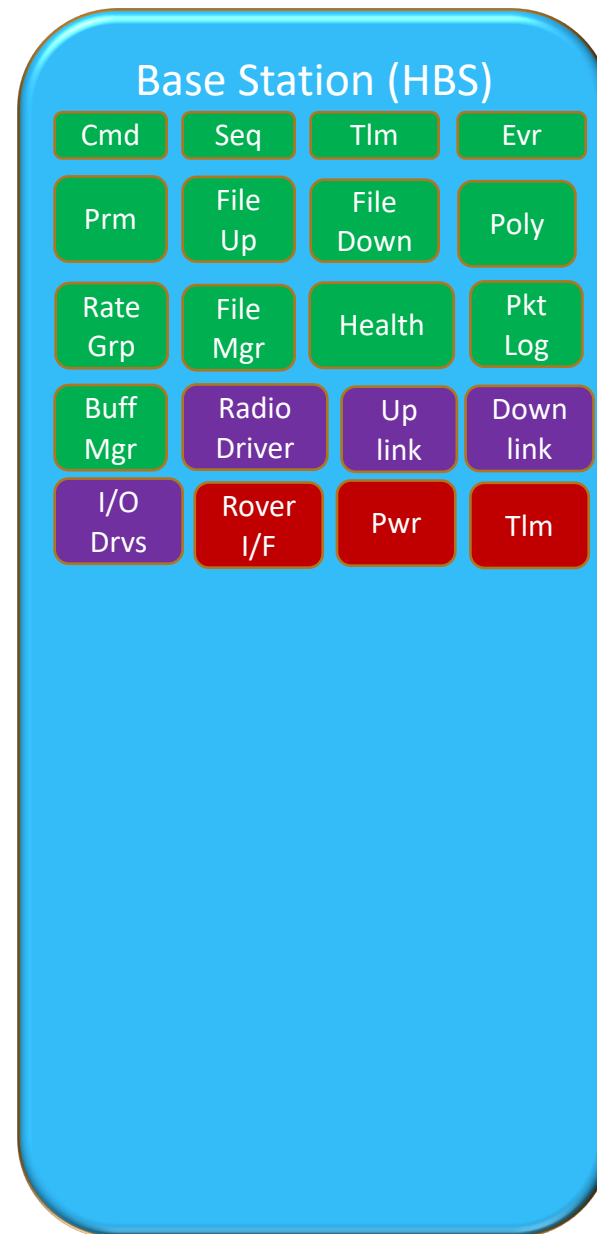
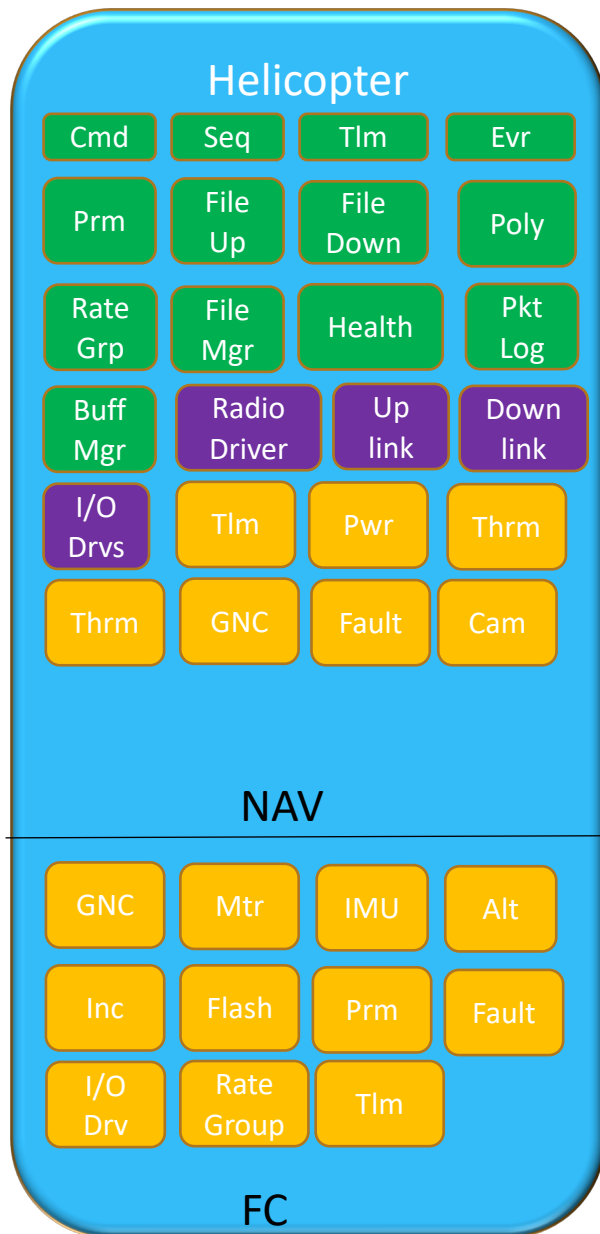
- Threading model
 - Telem, Cmds, Parameters
- Message Queues
- Ground interfaces
 - Sync, Async, Guarded



F Prime Development Process

- Define components and interfaces in XML
 - Code generation for boiler-plate code
 - Tasks, messages, commands and telemetry
- Developer writes C++ derived classes to implement component logic
- Generate unit test code to test component
 - Code generator generates component test harness
- Software lead assembles components into the topology
- System can be run with an included ground system with a python test API

Mars Helicopter Flight Software Components



Summary

- The use of COTS hardware allowed quick implementation of a compact, powerful avionics packet with a backstop of reliable flight parts
- We were able to achieve impressive performance that would not have been possible with conventional flight hardware
- The use of F Prime allowed us to leverage work done by other projects to mature the core components of the system
- The flexibility of F Prime allowed us support a number of venues and functions
- F Prime is available as open source
 - You too can fly code flown by the helicopter!
- Questions?

References

- BAE RAD750 - <https://bit.ly/2JFgM6T>
- Texas Instruments TMS570 - <https://bit.ly/2Yqhn02>
- Garmin Altimeter - <https://bit.ly/2xmh1M8>
- Bosch IMU - <https://bit.ly/2wBD2bn>
- JPL F Prime Framework - <https://bit.ly/2XD31fl>
- Mars Helicopter - <https://go.nasa.gov/2lybl3e>